

AIRBUS

A330

TEMPORARY REVISION

TR811 issue 1

A330NEO - HPV CLIP FAILURE MITIGATION

AIRPLANE FLIGHT MANUAL

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TR811 Issue 1 HPV CLIP FAILURE MITIGATION
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Ident.: TDU / APPRO-TR-00025892.0001001 / 25 AUG 22 Criteria: (330-800 or 330-900) Impacted DU: NONE	EASA APPROVED
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APPROVAL REFERENCE
APPROVED BY: EASA Approval date: 25 AUG 22 Approval reference: 10079993

Do not remove this Temporary Revision until instructed to do so.

Reason for issue: This Temporary Revision (TR) provides the limitation and procedures to mitigate the risk of a non isolated overpressure or overtemperature in the case of an excessive internal leak of the engine bleed High Pressure (HP) valve.

Applicable to: All A330neo aircraft.

This Temporary Revision is made up of the following Temporary Documentary Units:

APPRO-TR-00025892.0001001/25 AUG 22	ABN-36-00016652.0002001/25 AUG 22
LIM-36-00025899.0001001/25 AUG 22	NORM-PFLT-00025894.0001001/25 AUG 22
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BLEED SYSTEM

Ident.: **TDU / LIM-36-00025899.0001001 / 25 AUG 22** EASA APPROVED
Criteria: (330-800 or 330-900)
Impacted DU: NONE
Belongs to TR811 Issue 1

If a takeoff or go-around is performed with packs OFF or engine bleed OFF or APU bleed ON, a maintenance check of the conditions of the engine bleed HP valve seals after the flight is required in accordance with AOT Ref. A36L009-22.
For the following flight, takeoff is prohibited if the maintenance check is not performed.
This limitation does not apply in the case of dispatch under MMEL with pack(s) OFF or engine bleed OFF for the affected side(s). Refer to the associated MMEL item for the appropriate dispatch conditions.

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AIR - ENG BLEED NOT CLSD

Ident.: TDU / ABN-36-00025893.0001001 / 25 AUG 22 EASA APPROVED
Criteria: (330-800 or 330-900)
Impacted DU: NONE
Belongs to TR811 Issue 1

- **On ground:**
Return to the gate and request appropriate maintenance actions.

AIR - ENG BLEED FAULT

Ident.: TDU / ABN-36-00016652.0002001 / 25 AUG 22

EASA APPROVED

Criteria: (330-800 or 330-900)

Impacted DU: 00005117

Belongs to TR811 Issue 1

- **In the case of engine bleed overtemperature and the associated engine bleed valve failed open:**
 - **On ground:**

Return to the gate and request appropriate maintenance actions.
 - **In Flight:**

Turn off affected side engine bleed.

 - **If engine 1 bleed affected:**

Turn off APU bleed

Close crossbleed valve.

Turn off affected side pack.

Turn off wing anti-ice.

Note: Wing anti-ice is inoperative. Refer to ABN-30 A.ICE - WAI SYS FAULT or OFF.
- **In all other cases:**
 - **On ground:**

Return to the gate and request appropriate maintenance actions.
 - **In Flight:**

Turn off affected side engine bleed.

Open crossbleed valve.

 - **If wing anti-ice off:**

Set pack flow to low.

Set forward cargo cooling to off (if installed).
 - **If wing anti-ice on, or engine failure:**

Turn off affected side pack.

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BLEED SYSTEM

Ident.: TDU / NORM-PFLT-00025894.0001001 / 25 AUG 22 Criteria: (330-800 or 330-900) Impacted DU: NONE Belongs to TR811 Issue 1	<u>EASA APPROVED</u>
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- **Before each flight:**
Check in the Logbook the recorded packs and Engine/APU bleed configurations at takeoff and go-around (if performed) of the previous flight.
- **If the takeoff or go-around was performed with packs OFF or engine bleed OFF or APU bleed ON:**
Check that a maintenance action was done.
In case no maintenance action done, takeoff is not permitted.

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BLEED SYSTEM

Ident.: **TDU / NORM-ALDG-00025895.0001001 / 25 AUG 22** EASA APPROVED
Criteria: (330-800 or 330-900)
Impacted DU: NONE
Belongs to TR811 Issue 1

- **After each flight:**
Record in the Logbook the packs and Engine/APU bleed configurations at takeoff and go-around (if performed).
If the takeoff or go-around was performed with packs OFF or engine bleed OFF or APU bleed ON, maintenance action is due before the following flight.

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